

# REPORT DOCUMENTATION PAGE

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MEMORANDUM FOR PRS (Contractor/In-House Publication)

FROM: PROI (TI) (STINFO)

24 Oct 2000

SUBJECT: Authorization for Release of Technical Information, Control Number: **AFRL-PR-ED-TP-2000-205**  
Suri, Suresh; Tinnierllo, M. & Marcischak, J. (ERC), "Synthesis and Screening of Advanced Hydrocarbon Fuels"

**2000 USAF High Energy Density Matter (HEDM) Contractors Conference**  
**(Park City, UT, 24-26 Oct 2000) (Deadline: PAST)**

**(Statement A)**

1. This request has been reviewed by the Foreign Disclosure Office for: a.) appropriateness of distribution statement, b.) military/national critical technology, c.) export controls or distribution restrictions, d.) appropriateness for release to a foreign nation, and e.) technical sensitivity and/or economic sensitivity.  
Comments: \_\_\_\_\_  
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3. This request has been reviewed by the STINFO for: a.) changes if approved as amended, b.) appropriateness of distribution statement, c.) military/national critical technology, d.) economic sensitivity, e.) parallel review completed if required, and f.) format and completion of meeting clearance form if required  
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APPROVED/APPROVED AS AMENDED/DISAPPROVED

\_\_\_\_\_  
PHILIP A. KESSEL  
Technical Advisor

\_\_\_\_\_  
Date

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# Synthesis & Screening of Advanced Hydrocarbon Fuels

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# Presentation Outline

- **Goal**
  - HEDM program
  - NASA program
  - IHPRPT program ( propellant perspective)
- **Criteria for fuel selection**
- **Approach**
- **Results**
- **Accomplishments (FY-2000)**
- **Planned Efforts (FY-2001)**



# HEDM Goal

- To Develop fuels with increased Isp over  
LOX/RP-1
  - LOX/RP-1 (Calculated Isp) = 300 sec
  - LOX/RP-1 (Delivered Isp) = 263 sec

**Determined at sea level and 1000 psi chamber  
pressure**



# IHPRPT GOAL (Propellant Contribution)

## To Meet IHPRPT Phase II and Phase III Objective

Phase	Time	Improvement Over SOTA* Isp (del)
II	2005	+ 5 Sec
III	2010	+ 11 Sec

\*SOTA: LOX/RP-1 Propellant Isp(del) = 263 Sec.  
Isp (calc) = 300 Sec.



# NASA Goal

- **FY-1999**
  - Deliver three advanced hydrocarbon fuel in 8-10 lb quantity.
    - Quadricyclane
    - 1,7-Octadiyne
    - Bicyclopropylidene
- **FY-2000**
  - Screen four hydrocarbons for their physical and hazardous properties.



# Criteria for Fuel Selection

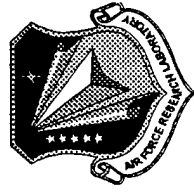
- Predicts better performance (Isp) over LOX/RP-1 system
- Most desirable physical properties
  - Lower vapor pressure compared to RP-1
  - Higher density ( $\geq$  RP-1 = 0.801 g/ml)
  - Freezing point ( $\leq -10$  °C; RP-1 = -41.4 °C)
  - Boiling point  $\geq$  B. P. Of RP-1
- Storable
- Compatible with the current system





# Approach

- Structural requirements
- Survey of energetic hydrocarbons
- Selection of hydrocarbons based on improved theoretical performance
- Synthesis of target hydrocarbons at bench scale.
  - *Easy preparation, cost effective and safe*
- Translate bench-scale synthesis to pilot scale.



# Heat of Formation of Saturated Hydrocarbons

Compound	Structure	$\Delta H_f$ (Obs)
Ethane	$\text{CH}_3\text{CH}_3$	-20.04
Propane	$\text{CH}_3\text{CH}_2\text{CH}_3$	-25.02
Butane	$\text{CH}_3(\text{CH}_2)_2\text{CH}_3$	-30.03
Pentane	$\text{CH}_3(\text{CH}_2)_3\text{CH}_3$	-35.08
	$\Delta H_f/\text{added CH}_2 = \sim -5 \text{ Kcal/mole}$	



# Heat of Formation of Unsaturated Hydrocarbons

Compound	Structure	$\Delta H_f(\text{Obs})$
Ethylene	$\text{CH}_2=\text{CH}_2$	+12.5
1,3-Butadiene	$\text{CH}_2=\text{CH}-\text{CH}=\text{CH}_2$	+26.11
	$\Delta H_f/\text{C} = \sim +6.25 \text{ Kcal/mole}$	
Acetylene	$\text{HC}\equiv\text{CH}$	+54.36
	$\Delta H_f/\text{C} = \sim +27.1 \text{ Kcal/mole}$	



## Structural Requirement for High Energy Contents (Cont..)

- The energy content is also increased by incorporating strain in the molecule

	$\Delta H_f$
– Ring compound	
– Cyclopropane	+ 12.73 kcal/mole
– Cyclobutane	+ 6.78 kcal/mole
– Cyclopentane	- 18.44 kcal/mole



# **Structural Requirements For High Energy Contents (Summary)**

**Incorporation of small ring (strain) and  
unsaturation in a molecule increases its  
energy contents**



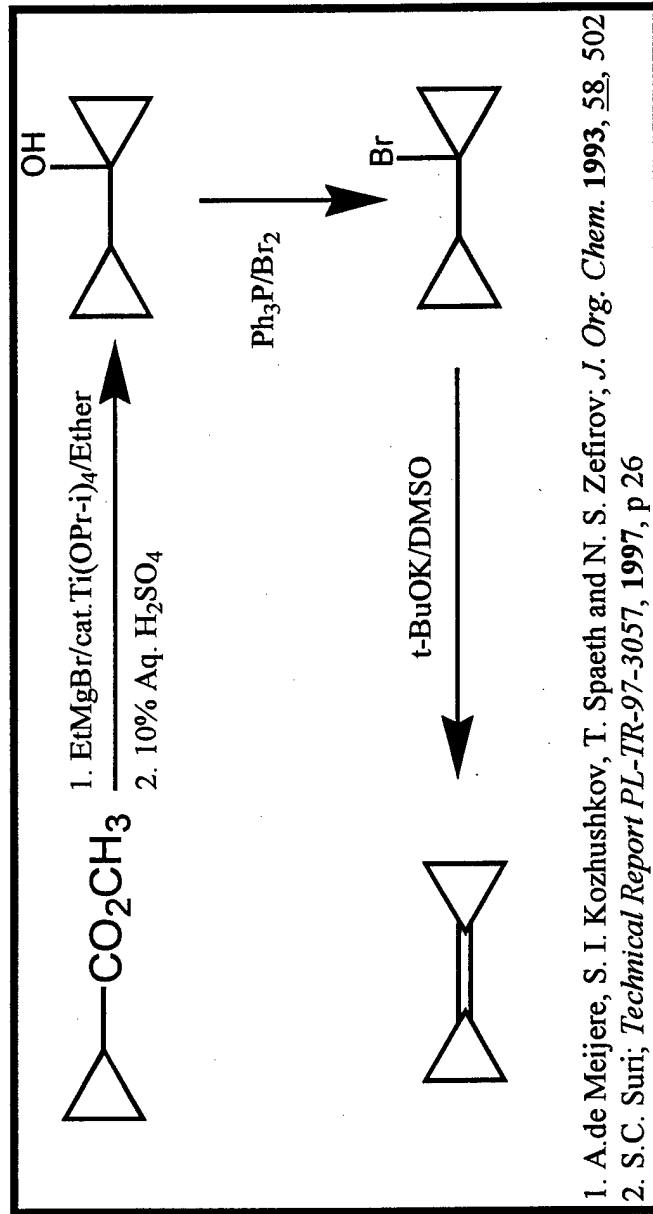
# Performance Comparison of Energetic Hydrocarbons (Theoretical)

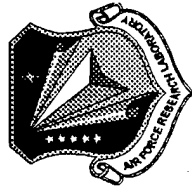
Hydrocarbons	H/C ratio	Density (g/ml)	Calc. $\Delta H_f$ (Kcal/mole)	Calc. Isp (sec)
RP-1	1.9	0.80	-5.76	300.0
Quad	1.14	0.98	72.2	307.0
BCP	1.33	0.85	76.1	312.5
AFRL-1	1.2	0.77	64.0	311.3
AFRL-2	1.25	0.87	73.4	307.2
AFRL-3	1.0	0.93	123.6	307.2
AFRL-4	1.0	-	129.6	321.4
AFRL-5	1.33	0.80	56.3	308.7



# Results

## Synthetic Sequence of BCP





# Characterization of BCP

## Physical properties

B.P. = 101 °C

M.P. = -12 °C

F.P. = -6.4 °C

Density = 0.8454 g/ml

$\Delta H_f$  (exp.) = 67.4 kcal/mole

$\Delta H_f$  (calc.) = 76.1 kcal/mole

## Hazardous properties

Zero card gap (negative)

Drop test > 200 kg/cm

Friction test 133 newton

## Toxicity

(Inhalation LC50)

1.95 mg/L

## Adiabatic Compression(psi)

3000

Neg.





# Is BCP Hypergolic?

- **Qualitative Test**

- BCP is found to be hypergolic using nitrogen tetroxide (NTO). Spontaneous reaction with visible flame.
- Hypergolic with inhibited red fuming nitric acid (IRFNA) as oxidizers. (Darren M. Thompson, U.S. Army missile command).

- **Ignition Delay**

- The work is in progress under SBIR phase-1 with TDA Research, Inc.



# Synthesis of AFRL-1

- Two steps synthesis
- Involves readily available materials
- Yield in both steps is  $> 90\%$



# Characterization of AFRL-1

## Physical Properties

B.P. = 52- 55 °C

Density = 0.77 g/ml

$\Delta H_f$  (Exp.)= 67.4 Kcal/mole

$\Delta H_f$  (Calc.)= 64.0 Kcal/mole

## Hazardous Properties

“0” card gap (Negative)

Liq. Impact test > 200 Kg-cm

Friction Test 78 Newtons

## Adiabatic Compression (psi)

3000

Neg.



# Synthesis of AFRL-3

- One step synthesis from AFRL-1.
- Requires oxidative coupling of AFRL-1.
- Yield is 92 %.



# Characterization of AFRL-3

## Physical Properties

B.P. = 102 °C

M.P. = -13 °C

Density = 0.93 g/ml

$\Delta H_f$  (Calc.) = 123.6 kcal/mole

$\Delta H_f$  (Exp.) = 117.0 kcal/mole

## Hazardous Properties

- “0” card gap (negative)
- Liq Impact test <20 kg-cm
- Friction Test = 64.8 Newton

## Adiabatic Compression(psi)

500	Neg.
2000	Neg.
3000	Neg.



# Synthesis of AFRL-5

- Higher homologue of AFRL-1
- Two step synthesis
- Yield in both steps is greater than 90 %



# Characterization of AFRL-5

## Physical properties

B.P. = 78 °C

M.P. = -92.8 °C

Density = 0.7957 g/ml

$\Delta H_f$  (Exp.) = 50.39  
kcal/mole

$\Delta H_f$  (Calc.) = 56.3 kcal/mole

## Hazardous properties

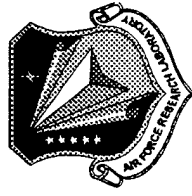
“0”card gap (TBD)

Liq. Impact Test > 200 kg-cm

Friction Test = 43.12 newton

Adiabatic Compression (psi)

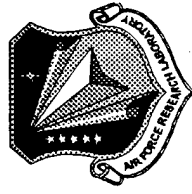
3000	Neg.
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## Accomplishments (FY 00)

- Delivered four hydrocarbons to NASA/Marshall.
  - Cyclopropyl acetylene (AFRL-1).
  - Bicyclopropylidene
  - Quadricyclane
  - 1,7-Octadiyne
- Synthesized two advanced hydrocarbons (AFRL-1 & AFRL-3) at bench-scale level.
- 200 gm of AFRL-3 was synthesized in the laboratory.





# Planned Efforts of Fiscal Year 2001 (Technical)

- To continue exploring bench scale synthesis of advanced hydrocarbon (AFRL-4).
- Evaluate physical & hazardous properties of AFRL-4 & AFRL-2.



# Alliances

- **Industry**
  - Boeing
  - TRW
  - Kistler
  - Aerojet
- **NASA**
  - Marshall
  - Glenn
- **DOD**
  - Navy- China Lake
  - Army- Huntsville



# Team Efforts

## Research

- Suresh C. Suri  
Michael Tinnirello  
Jacob Marcischak

## Theoretical Efforts

- Jeffrey Mills

## Physical Properties

- Paul Jones, JoAnne Larue,  
Jeff Yinn

## Hazardous Properties

- Tommy W. Hawkins,  
Adam Brand, Milton  
Mckay, Ismail Ismail



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